

**CORRELATION BETWEEN NO-SLIP AND SLIP
BOUNDARY CONDITIONS ASSOCIATED WITH
A TWO-DIMENSIONAL NAVIER-STOKES
FLOWS IN A PLANE DIFFUSER**

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Abstract

We examine the viscous effects of slip boundary conditions for the model describing two-dimensional Navier-Stokes flows in a plane diffuser. It is shown that the velocity profile is related to a half period shifted Weierstrass function with two parameters. This allows to approximate the explicit solution by a Taylor series expansion with two new micro- parameters, that can be measured in physical experiments. It is shown that the assumption for no-slip boundary conditions is stable in the sense that a small perturbation of the boundary values result in a small perturbation in the solutions.

1. Introduction

The governing equations, along with the appropriate constitutive relations describe fully the fluid flow within a given geometry. However, mathematical model cannot be solved unless we specify the boundary conditions for the problem. Physically, the need for boundary conditions is not surprising, because the boundary conditions provide information on the way the fluid interacts with its surroundings. Mathematically, this is expected because upon integration, the differential equations yield constants that are fixed using boundary conditions.

In this paper, we will consider the viscous effects of slip boundary conditions for the model describing two-dimensional Navier-Stokes flows in a plane diffuser. A diffuser is an engineering device used within internal flow systems widely used in industry. For example, ejectors used in refrigeration systems as entrainment and compression components or expanders, attract interest from engineering as a means of more efficient energy use ([1, 6, 19]). Commonly, such complex flows in a diffuser are encountered in turbomachines between a compressor and a combustor or at the exit of a turbine, as shown in Figure 1. Diffusing passages may also be encountered in small-scale devices such as fluidic actuators and Micro-Electro-Mechanical Systems.

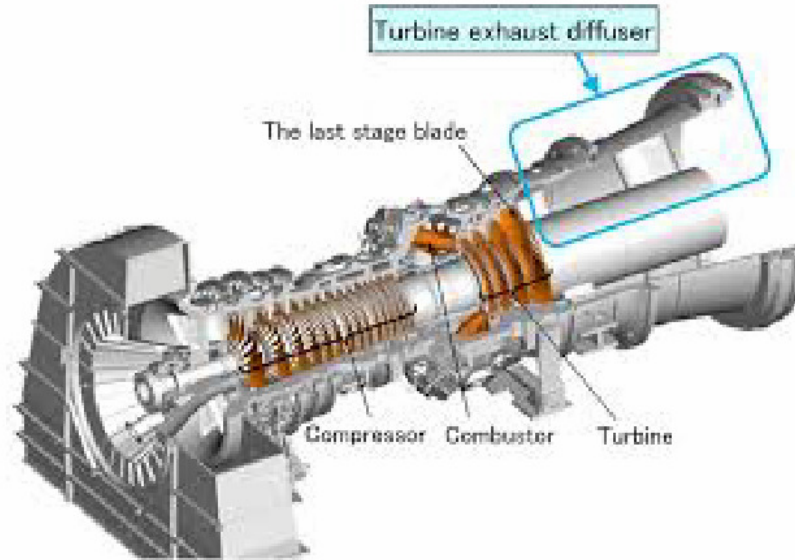


Figure 1. Turbine exhaust diffuser. Source [10].

It is known that in nozzles the static pressure decreases in the flow direction (favourable pressure gradient) [3, 11]. However, fluid particle flowing through a diffuser experiences an increasing static pressure (adverse pressure gradient) resulting to complex fluid phenomena, including flow separation and unsteadiness to transitory stall and violent flow-excited static pressure fluctuations ([4, 20]). Such flows are hard to predict quantitatively, although they are qualitatively understood [5, 18].

In the vast majority of problems, we have fluid flowing along solid boundaries. Consistent with continuum theory, the *no-slip* boundary condition states that there is no relative motion between fluid particles and the solid boundaries with which they are in contact. That is,

$$V_{\text{fluid}} = V_{\text{wall}}. \quad (1)$$

No slip condition is the reason for formation of boundary layer in a fluid flow [13]. If the wall is not moving, then the fluid velocity at the wall is

$$V_{\text{fluid}} = 0. \quad (2)$$

Basically the molecules of the fluid crash into the molecules of the wall and get stopped. Then they dribble out and get crashed into by faster molecules farther out in the flow.

For example, all gases experience some slip at the wall where typically the slip velocity V (difference between tangential velocities of wall and gas) is given by a condition like

$$V = - a_1 s_{tn} - a_2 q_t, \quad (3)$$

where s_{tn} is the tangential-normal element of pressure tensor, q_t is tangential heat flux, and a_1 , a_2 are coefficients that depend on the state of the gas. The first term describes classical slip, the second term describes what is known as thermal transpiration: a flow driven by a temperature gradient along the wall ([16]).

Despite its apparent simplicity, the no-slip boundary condition cannot be derived from first principles and could, in theory, be violated. In addition, it leads to some physical inconsistencies, that are not resolved completely. For instance, the no-slip boundary condition cannot explain the motion of a liquid interface in contact with a solid boundaries; according to this condition, the liquid interface in a partially filled glass must remain stationary with respect to the glass when the glass is moved. At the microscopic level at least, we know that this is not the case. Consider also the flow near an exist of a pipe. According to the no-slip condition, the velocity of fluid particles near the exit is mathematically singular as the velocity changes from zero to a finite value.

It has been shown in [15] that the physical inconsistencies can be removed mathematically by assuming that the fluid particles in contact with solid surface slip. This means that fluid particles in contact with the wall can slide, resulting in an apparent fluid velocity. As a matter of fact, there is a strong evidence (see, e.g., [14, 17]) that this occurs in many non-Newtonian fluids. In such cases, the slip velocity V_s is assumed to be proportional to the shear along the wall,

$$V_s = \lambda \tau_w, \quad (4)$$

where λ is a proportionality constant to be determined experimentally and τ_w is shear along the wall (see also [9]). However, for the most Newtonian fluids there are not many experimental evidence to support the presence of a slip [12]. In the present work, we consider a special case of a diffuser for which we study the relation between V_s and the shear. We show that no-slip boundary conditions can be considered as a limit case, $\lambda \rightarrow 0$, of the slip boundary conditions and we study the effect on the flow for different ranges of λ .

2. Mathematical Model and Problem Geometry

In two dimensions, we introduce polar coordinates $x = r \cos \theta$, $y = r \sin \theta$ and write two-dimensional Navier-Stokes equations describing a fully developed flow within a plane diffuser, as a two dimensional channel with the line source at the origin, as shown in Figure 2

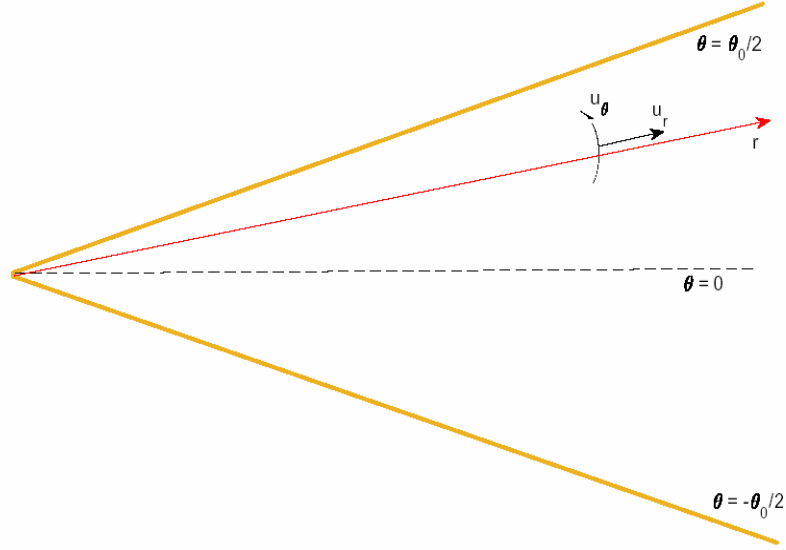


Figure 2. Problem geometry.

$$u_r \frac{\partial u_r}{\partial r} + \frac{u_\theta}{r} \frac{\partial u_r}{\partial \theta} - \frac{u_\theta^2}{r} = -\frac{1}{\rho} \frac{\partial p}{\partial r} + \nu \left(\nabla^2 u_r - \frac{2}{r^2} \frac{\partial u_\theta}{\partial \theta} - \frac{u_r}{r^2} \right), \quad (5)$$

$$u_r \frac{\partial u_\theta}{\partial r} + \frac{u_\theta}{r} \frac{\partial u_\theta}{\partial \theta} + \frac{u_\theta u_r}{r} = -\frac{1}{\rho} \frac{1}{r} \frac{\partial p}{\partial \theta} + \nu \left(\nabla^2 u_\theta + \frac{2}{r^2} \frac{\partial u_r}{\partial \theta} - \frac{u_\theta}{r^2} \right), \quad (6)$$

$$\frac{\partial}{\partial r} (r u_r) + \frac{\partial u_\theta}{\partial \theta} = 0, \quad (7)$$

where u_r and u_θ are the radial and angular components of the velocity vector, ρ is a constant density, $p(r, \theta)$ is a pressure and

$$\nabla^2 = \frac{\partial^2}{\partial r^2} + \frac{1}{r} \frac{\partial}{\partial r} + \frac{1}{r^2} \frac{\partial^2}{\partial \theta^2}. \quad (8)$$

From (7), we can introduce the stream function $\psi(r, \theta)$ via

$$u_r = \frac{1}{r} \frac{\partial \psi}{\partial \theta}, \quad u_\theta = -\frac{\partial \psi}{\partial r}. \quad (9)$$

3. Basic Flow using Stream Function

The basic flow is assumed to be purely radial emerging from a line source, so we can look for particular solution for the problem in the form

$$u_r^0 = u_r^0(r, \theta), \quad u_\theta^0 = 0, \quad p^0 = p^0(r, \theta). \quad (10)$$

Then, as follows from Equation (5)-(6), the basic flow (10) satisfies

$$u_r^0 \frac{\partial u_r^0}{\partial r} = -\frac{1}{\rho} \frac{\partial p^0}{\partial r} + v \left(\frac{\partial^2 u_r^0}{\partial r^2} + \frac{1}{r} \frac{\partial u_r^0}{\partial r} + \frac{1}{r^2} \frac{\partial^2 u_r^0}{\partial \theta^2} - \frac{u_r^0}{r^2} \right), \quad (11)$$

$$0 = -\frac{1}{\rho} \frac{1}{r} \frac{\partial p^0}{\partial \theta} + v \left(\frac{2}{r^2} \frac{\partial u_r^0}{\partial \theta} \right). \quad (12)$$

Also, it follows from (7) that ψ is a function of θ only.

Hereafter the index zero is omitted. We next write the model (11)-(12) in terms of ψ as follows:

$$\frac{1}{r^3} \left(\frac{\partial \psi}{\partial \theta} \right)^2 = \frac{1}{\rho} \frac{\partial p}{\partial r} - v \left(\frac{1}{r^3} \frac{\partial^3 \psi}{\partial \theta^3} \right), \quad (13)$$

$$0 = -\frac{1}{\rho r} \frac{\partial p}{\partial \theta} + v \left(\frac{2}{r^3} \frac{\partial^2 \psi}{\partial \theta^2} \right). \quad (14)$$

We note that the pressure p can be determined from Equation (14) as

$$\frac{1}{\rho} p(\theta, r) = \frac{2v}{r^2} \frac{\partial \psi}{\partial \theta} + \alpha(r), \quad (15)$$

where α is an arbitrary function.

4. Derivation of the Main Equation

We rewrite the Equation (13), in which p is given by (15)

$$\frac{\partial^3 \psi}{\partial \theta^3} + 4 \frac{\partial \psi}{\partial \theta} + \frac{1}{v} \left(\frac{\partial \psi}{\partial \theta} \right)^2 = \frac{r^3}{v} \frac{d\alpha}{dr}. \quad (16)$$

The left part of (16) depends only on θ whereas the right side depends only on r . Hence, the both sides are equal to a constant, say c_1 . Then the right side of (16) yields

$$\alpha(r) = -\frac{c_1 v}{2r^2} + \frac{p_0}{\rho}, \quad (17)$$

where $p_0 = \text{const}$. By substituting $\alpha(r)$ given by (17) into (15), we obtain the following expression for the pressure:

$$p(\theta, r) = \frac{\nu \rho}{r^2} \left[2 \frac{\partial \mathcal{P}}{\partial \theta} - \frac{c_1}{2\rho} \right] + p_0. \quad (18)$$

Next we modify the main equation and the boundary conditions. We denote

$$U(\theta) = \frac{\partial \mathcal{P}}{\partial \theta}. \quad (19)$$

Then in terms of U , the Equation (16) is written as

$$\frac{d^2 U}{d\theta^2} + 4U + \frac{1}{\nu} U^2 = c_1, \quad (20)$$

where c_1 is an arbitrary constant. Multiplying (20) by dU and integrating we get the first order equation

$$\left(\frac{dU}{d\theta} \right)^2 = -\frac{2}{3\nu} U^3 - 4U^2 + 2c_1 U - 2c_2, \quad (21)$$

where c_2 is a new constant.

We note that the variables in Equation (21) can be separated, which provides us with the relation between U and θ in the form

$$\theta = \int_{U_1}^U \frac{d\xi}{\pm \sqrt{f(\xi)}}, \quad (22)$$

where

$$f(\xi) = -\frac{2}{3\nu} \xi^3 - 4\xi^2 + 2c_1\xi - 2c_2. \quad (23)$$

The form (22) suggests that we can find the exact solution of the general equation (21) by the substitution

$$U(\theta) = -6\nu \left(\frac{1}{3} + y(\theta) \right), \quad (24)$$

where y is solution of the equation

$$y'' = 6y^2 - \frac{g_2}{2} \quad (25)$$

for a real constant g_2 . The general solution of the above equation is the Weierstrass elliptic P function, denoted \wp , [7]. The function $\wp(\theta, g_2, g_3)$ is a well known function that posses interesting properties and is the solution for our BVP problem. The first parameter g_2 appears in Equation (25) whereas the second parameter g_3 appears after integration of Equation (25). These two parameters g_2, g_3 has to be determined by the boundary conditions. Next we provide brief overview of the function \wp and it's application to our model.

5. Application of the Weierstrass Elliptic Function

The Weierstrass elliptic function \wp is well studied and has many applications in different branches of pure and applied mathematics, a good source for reference is [7]. The function \wp is doubly periodic with complex periods w_1, w_2 and can be defined as the solution of the equation

$$\wp'' = 6\wp^2 - \frac{g_2}{2}, \quad (26)$$

or the equivalent first order DE

$$\wp'^2 = 4\wp^3(z) - g_2\wp(z) - g_3, \quad (27)$$

where g_2 and g_3 are real valued constants.

The function \wp is even, $\wp(-z) = \wp(z)$, with poles at a lattice generated by the periods including $z = 0$ and it has a Laurent expansion near the pole $z = 0$ of the form

$$\wp(z) = \frac{1}{z^2} + \frac{g_2}{20}z^2 + \frac{g_3}{28}z^4 + \sum_{k=4}^{\infty} c_k z^{2k}, \quad (28)$$

where

$$c_k = \frac{3}{(2k+1)(k-3)} \sum_{m=2}^{k-2} c_m c_{k-m}, \quad (29)$$

with $c_2 = \frac{g_2}{20}$ and $c_3 = \frac{g_3}{28}$.

We are looking for a real valued function U and hence real valued $\wp(\theta)$ for real θ . From [8] for w_1, w_2 real and hence g_2, g_3 are also real, the general solution of the equation

$$y'^2 = 4y^3 - g_2y - g_3 \quad (30)$$

is real valued function and it is either $\wp(\theta)$ or $\wp\left(\theta + \frac{w_1}{2}\right)$, where w_1 is one of the periods for the \wp function. We will show that the Equation (20) for the diffuser is solved by the second function for $\theta \in \left[-\frac{\theta_0}{2}, \frac{\theta_0}{2}\right]$ thus avoiding the poles of $\wp(\theta)$. If $\theta_0 < 2 \min(w_1, w_2)$ the function $\wp\left(\theta + \frac{w_1}{2}\right)$ has a single minimum and for θ away from the poles, it is analytic in θ, g_2, g_3 , which provides continuity of the solution with

respect to the boundary conditions. Small variations from zero initial conditions result in small variations in U so this might justify the zero conditions in some cases. On the other hand since \wp is defined beyond the limits of the interval then (27) provides the relation

$$\frac{1}{36v^2} U'^2\left(\frac{\theta_0}{2}\right) = -4\left(\frac{U_0}{6v} + \frac{1}{3}\right)^3 + g_2\left(\frac{U_0}{6v} + \frac{1}{3}\right) - g_3. \quad (31)$$

Denoting by U_0 the boundary value $U\left(\frac{\theta_0}{2}\right)$ we see that zero initial

condition, i.e., $U_0 = 0$ corresponds to (31) as $U'_0 = 36v^2\left(\frac{4}{27} - \frac{g_2}{3} - g_3\right)$

and is zero only if $\frac{4}{27} - \frac{g_2}{3} - g_3 = 0$ but then $U \leq 0$ and is not a feasible solution. We can conclude that $U_0 = \mu\lambda U'_0$ is not possible for $U_0 = 0$ unless $\mu = 0$. It is interesting to notice that if $U_0 > 0$ than the cubic relation (31) provides a nonlinear relation between the initial values of U and U' which is similar in nature to (4). We discuss that fact in Section 8.

The amount of fluid for any r is conserved which results in the conserving condition

$$\int_{-\frac{\theta_0}{2}}^{\frac{\theta_0}{2}} U(\theta) d\theta = 2 \int_0^{\frac{\theta_0}{2}} U(\theta) d\theta. \quad (32)$$

It can be expressed in terms of Weierstrass ζ function, $\frac{d\zeta}{d\theta} = -\wp(\theta)$. The Weierstrass elliptic functions are implemented in Mathematica and we provide an example.

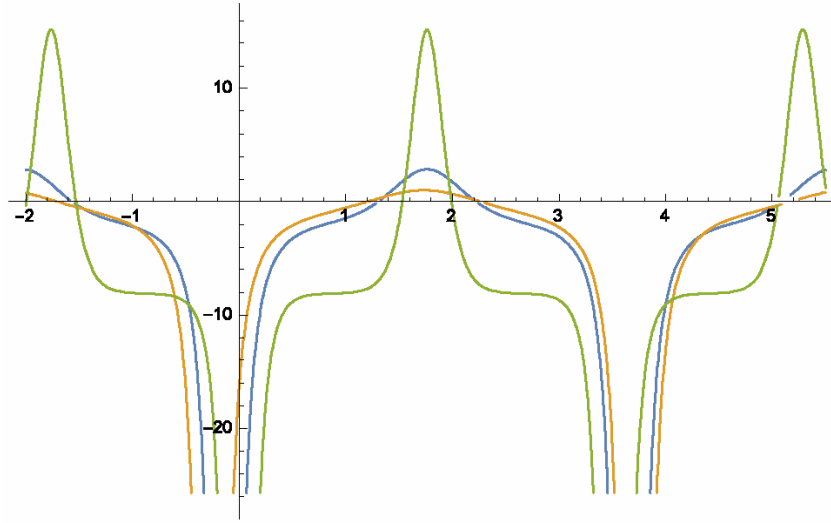


Figure 3. Three graphs for \wp shifted to have the maximum coinciding.

Figure 3 shows three graphs for \wp shifted to have the maximum coinciding. These three graphs represent

$$- \wp(x + 0.13, 29, -37) - 1/3,$$

$$- \wp(x + 0.25, 2, -7) - 1/3,$$

$$- \wp(x, 724.1, -3750) - 1/3.$$

All of the functions are analytic in a neighbourhood around the maximum value but have irremovable singularities outside finite intervals centered at the maximum points. Another observation is that the profile of \wp depending on the parameters g_2, g_3 can vary a lot with the variation of the parameters. In the next section we conduct numerical study of the solutions of the BVP.

6. Numerical Study

The Weierstrass function $\wp(\theta, g_2, g_3)$ has different representations. Computationally, the elliptic theta functions provide better convergence than the Laurent series but for our problem we take a different approach. The parameterization by g_2, g_3 is not appropriate for accessing values of the function. The Laurent series expansion is slowly convergent near the poles but using the fact that the function is analytic on the domains of our interest we mimic the construction of \wp for the diffuser problem. We consider two macro-parameters that could be easily measured during an experiment, the maximum velocity of the flow, E , and since the amount of fluid is con-served for any r we consider it as a parameter, say Q . The following problem is discussed next.

Problem. Approximate the solutions of

$$u''(\theta) = -\frac{1}{v}u^2(\theta) - 4u(\theta) + s, \quad (33)$$

with the constrain

$$\int_{-\frac{\theta_0}{2}}^{\frac{\theta_0}{2}} u(\theta) d\theta = Q > 0, \quad (34)$$

where v is the viscosity of the fluid, θ_0, Q are positive constants, and s is a real parameter.

The solution of the DE (33) is $\wp(\theta + \frac{w_1}{2})$, which is analytic and even function on $[-\frac{\theta_0}{2}, \frac{\theta_0}{2}]$, hence it has the Maclaurin series expansion

$$u(\theta) = \sum_{k=0}^{\infty} \alpha_k \theta^{2k}, \quad (35)$$

with coefficients

$$a_k = \frac{u^{(2k)}(0)}{(2k)!}. \quad (36)$$

We introduce two new parameters: the maximum value $U(0) = E$ and s from (33). Using the DE we can find a recurrence relation for a_k . For $k = 0$ we have $a_0 = u(0) = E$, for $k = 1$ from the equation

$$a_1 = \frac{u''(0)}{2} = -\frac{1}{2v} E^2 - 2E + \frac{s}{2}, \quad (37)$$

and for $k > 1$ by differentiating the DE $(2k - 2)$ -times we get

$$\begin{aligned} \frac{u^{(2k)}}{(2k)!} &= -\frac{1}{v(2k)!} (u^2)^{(2k-2)} - 4 \frac{u^{(2k-2)}}{(2k)!} \\ &= -\frac{1}{v(2k)!} \sum_{m=0}^{2k-2} \binom{2k-2}{m} u^{(2k-2-m)} u^{(m)} - 4 \frac{u^{(2k-2)}}{(2k)!}. \end{aligned} \quad (38)$$

Since $u^{(2m-1)}(0) = 0$ we have

$$\begin{aligned} \frac{u^{(2k)}(0)}{(2k)!} &= -\frac{1}{v(2k)(2k-1)} \sum_{m=0}^{k-1} \frac{u^{(2k-2-2m)}(0)}{(2k-2-2m)!} \frac{u^{(2m)}(0)}{(2m)!} \\ &\quad - \frac{4}{2k(2k-1)} \frac{u^{(2k-2)}(0)}{(2k-2)!}, \end{aligned} \quad (39)$$

and

$$a_k = -\frac{1}{2k(2k-1)} \left(\frac{1}{v} \sum_{m=0}^{k-1} a_{k-m-1} a_m + 4a_{k-1} \right). \quad (40)$$

The first seven terms are computed as follows:

$$\begin{aligned}
 a_0 &= E, \\
 a_1 &= -\frac{1}{2\nu} E^2 - 2E + \frac{s}{2}, \\
 a_2 &= \frac{-1}{12} \left(\frac{1}{\nu} 2a_0 a_1 + 4a_1 \right), \\
 a_3 &= \frac{-1}{30} \left(\frac{1}{\nu} (2a_0 a_2 + a_1^2) + 4a_2 \right), \\
 a_4 &= \frac{-1}{56} \left(\frac{1}{\nu} (2a_0 a_3 + 2a_2 a_1) + 4a_3 \right), \\
 a_5 &= \frac{-1}{90} \left(\frac{1}{\nu} (2a_0 a_4 + 2a_3 a_1 + a_2^2) + 4a_4 \right), \\
 a_6 &= \frac{-1}{132} \left(\frac{1}{\nu} (2a_0 a_5 + 2a_4 a_1 + 2a_3 a_2) + 4a_5 \right).
 \end{aligned}$$

Considering them as functions of s all terms a_1, \dots, a_6 are at most cubic. The conserve condition for $Q > 0$ is

$$\int_{-\frac{\theta_0}{2}}^{\frac{\theta_0}{2}} u(\theta) d\theta = Q. \quad (41)$$

This condition for the approximation $u(\theta) \approx \sum_{m=0}^6 a_m \theta^{2m}$ (which is a polynomial of degree 12 for $n = 6$) is

$$2qa_0 + \sum_{m=1}^n \frac{2q^{2m+1}}{2m+1} a_m = Q, \quad (42)$$

where $q = \frac{\theta_0}{2}$, and is a cubic equation in s . In general for a fixed maximum value E , the n -Taylor's polynomial is a polynomial for c and we can use numerical methods effectively to approximate s such that the solution satisfies the conserve condition.

Next we consider $x \in [-\frac{\pi}{4}, \frac{\pi}{4}]$ and study how the boundary values at $\pm \frac{\pi}{4}$ and Q change by varying E and ν .

Experiment 1. Fix $\nu = 1$, $Q = 5$, then for $E = 2, \dots, 6$ from (42) with $n = 9$ we estimate $s = [27.5, 23.5, 20.5, 19, 18.5]$. Next, using (35) we get the approximate solution shown in Figure 4.

The approximate values for $Q = [5.0656, 5.0338, 4.9880, 4.9910, 4.9874]$.

The solutions in Figure 4 show that the boundary conditions decrease smoothly to zero when E varies from 2 to 5, and become to negative for $E = 6$. In the case $E > 6$, the solution exists only for negative initial values. This might be considered corresponding to walls moving in opposite direction of the flow. Another observation is that for small values of E but $Q = 5$ the maximum velocity is along the walls.

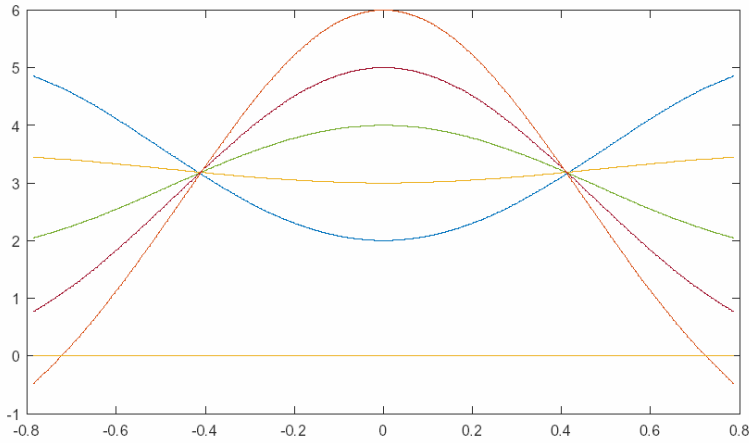


Figure 4. $Q = 5$ and $E = 2, \dots, 6$.

Experiment 2. Fix $\nu = 1$, $E = 10$ and let s from 52 to 86, skip 5. By using (35), we obtained the following solutions:

The conserve quantity is $Q = [7.8194, 8.3485, 8.8662, 9.3732, 9.8700, 10.3570, 10.8347]$.

According to the numerical results shown in Figure 5, for a fixed maximum value, $E = 10$, by varying s the conserve quantity varies smoothly and the boundary conditions vary from negative to positive.

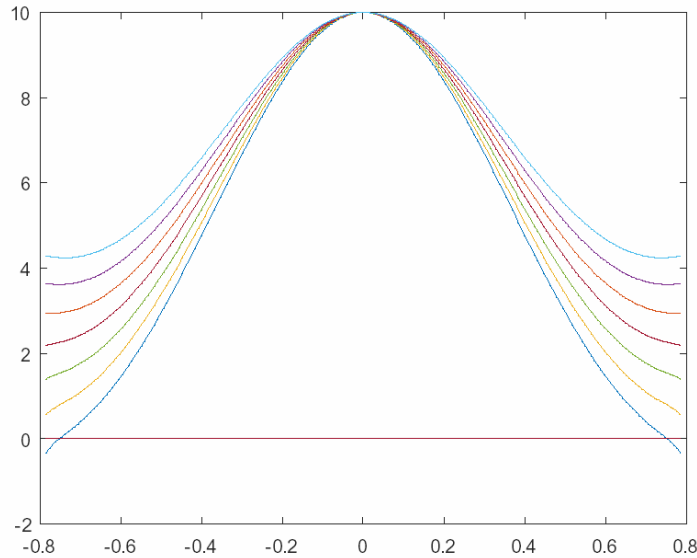


Figure 5. $E = 10$ and s varies from 52 to 86, skip $s = 5$.

Experiment 3. In the last experiment we study the effect when ν varies from 0.35 to 2.5 with a step 0.2. The maximum value is fixed as $E = 4$ and $s = 17$. Using the McLaurin series again we obtained the following solutions:

The conserve quantity is $Q = [2.3411, 3.4708, 4.0885, 4.4909, 4.7752, 4.9873, 5.1517, 5.2830, 5.3902, 5.4795, 5.5550]$.

Figure 6 shows that for $\nu = 0.35$ the initial conditions are negative and when ν increases they smoothly increase.

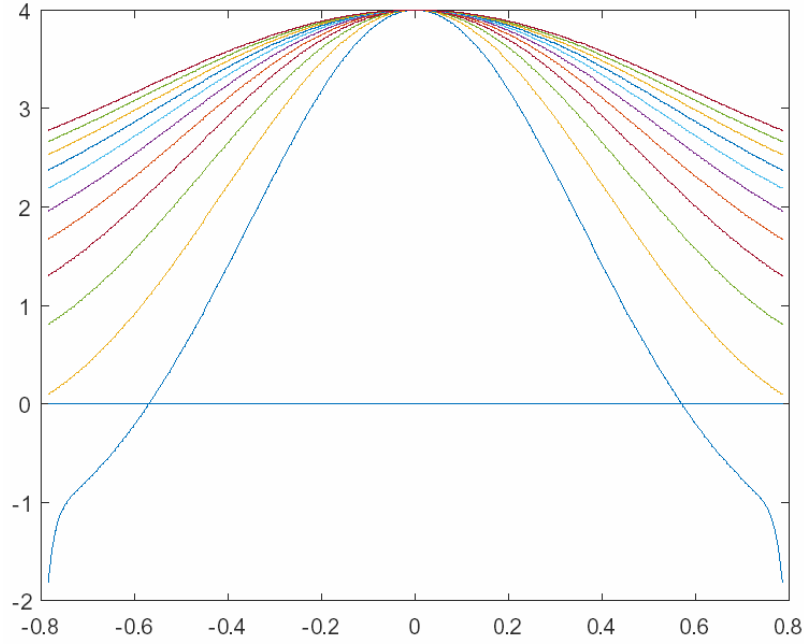


Figure 6. $E = 10$ and s varies from 52 to 86, skip $s = 5$.

7. Slip vs. No-Slip Boundary Conditions

In this section, we show that the no-slip boundary condition can be obtained as a uniform limit of the slip boundary condition. Following the notation in (4) we introduce

$$U_s = -\lambda\mu \frac{dU}{d\theta} \Big|_{\theta=\pm\frac{\theta_0}{2}}, \quad (43)$$

where U_s is the fluid velocity along the wall $\theta = \pm\theta_0/2$. For ease of notations we assume $\lambda = 1$ and then the slip boundary condition (4) becomes $U_s = -\mu U'_s$. When coupled with (33) it is equivalent to $U_s(\mu)$ being a positive solution of the cubic equation

$$\frac{1}{\mu^2} U_s^2 = f(U_s), \quad (44)$$

where f is introduced by (23).

The cubic polynomial $P(x, \mu) = \frac{2}{3\nu} x^3 + (4 + \frac{1}{\mu})x^2 - 2c_1x + 2c_2$ as a function of μ is uniformly decreasing for any x , indeed

$$\frac{\partial P}{\partial \mu} = -\frac{2}{\mu^3} x^2 < 0, \quad (45)$$

for $\mu > 0$. When $\mu \rightarrow \infty$, $P(x, \mu) \approx x^2$ with one positive and one negative zeros in a neighbourhood of $x = 0$, the third zero approaches $-\infty$. When $\mu \rightarrow 0$ $P(x, \mu) \rightarrow P(x, 0)$ and the zeros approach the zeros of $P(x, 0)$. Our interest is in the behaviour of $U_s(\mu)$ when $\mu \rightarrow 0$. From the above considerations, it follows that $U_s(\mu)$ is the greatest positive solution of the cubic equation $\mu P(x, \mu) = 0$ and $U_s(0) = 0$. By using the formulas for solutions of cubic equations (44), we can conclude that the positive zero $U_s(\mu) = 0 + O(\mu^2)$ when $\mu \rightarrow 0$. For any choice of c_1, c_2 and the power series expansion of U it follows that $U(\theta, \mu) \rightarrow U(\theta, 0)$ as $\mu \rightarrow 0$. An example with viscosity $\nu = 1$ and $c_1 = 35, c_2 = 17$ on the interval $[-\frac{\pi}{8}, \frac{\pi}{8}]$ is illustrated in Figure 6. Table 1 contains the values of μ and the corresponding $U_s(\mu)$ as well as the maximum velocity in each case.

Table 1

μ	0	0.03	0.05	0.07	0.09	0.15	0.21	0.26	0.36	0.47	8	800
$U_s(\mu)$	0	0.20	0.37	0.54	0.88	1.81	2.85	3.73	4.42	4.42	4.42	4.42
E	17.50	17.25	17.00	16.75	16.25	14.75	12.75	10.50	6.25	6.25	6.25	6.25

For small values of μ the boundary conditions are close to zero with maximum velocity in the velocity of 17. For larger values of $\mu > 0.36$ the solution U convergence very fast to the limit case $U(\theta, 0)$. We see that slip boundary conditions result in solutions that uniformly converge to the solution in the no-slip case when $\mu \rightarrow 0$, which is also illustrated in Figure 7.

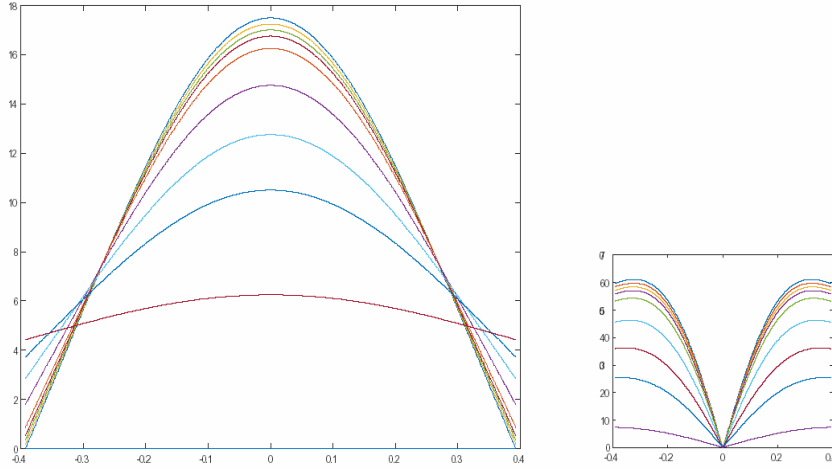


Figure 7. Slip vs no-slip BC: right graph is $U(\mu)$, left graph is $U_\theta(\mu)$.

8. Conclusion

We studied the correlation between no-slip and slip boundary conditions for Newtonian fluid for special case of a plane diffuser. The general solution of Equations (20) and (25) for the velocity is related to a half period shifted Weierstrass P function with parameters g_2, g_3 . The

explicit solution obtained is a well known function in many areas of mathematics but for our analysis we provided a Taylor series expansion centered at 0 with two new parameters, the maximum velocity E and the parameter s . In Section 7, we presented a numerical study of the flow described by two macro-parameters, the maximum flow and the conserved quantity Q . These two quantities can be easily measured in physical experiments. In a series of examples, we showed that the assumption for no-slip boundary conditions is stable in the sense that a small perturbation of the boundary values result in a small perturbation in the solutions.

In Section 8, we established that slip boundary conditions are uniform limit of no-slip boundary conditions and considered constructive method for solving the boundary value problem

$$\frac{d^2U}{d\theta^2} + 4U + \frac{1}{\nu}U^2 = c_1, \quad (46)$$

$$U_s = -\mu \frac{dU_s}{d\theta}. \quad (47)$$

Since ejector-based systems provide a potentially promising solution, particularly for moderate heating or cooling applications, refrigeration and air conditioning, the results presented in this work might be useful in physical experiments studying the fluid velocity and with slip or no-slip effect. A better understanding the correlation between slip and no-slip boundary conditions in a diffuser may improve the technology of ejector-based systems, which can recover low-grade energy available as waste heat in most industrial processes and use renewable energy or any other source at low cost.

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